

A new approach for precise orbit determination based on raw GNSS measurements

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Basic considerations

- Basic principle in geodesy:
 - Observations are directly used to estimate parameters
- In GNSS processing this principle is neglected
- Observations are combined beforehand to form derived observations
 - Linear combinations, differences, ...

Basic considerations

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Why not use them directly?

Principles for precise orbit determination

- Use all available observations as they are observed
 - Code and phase
- No forming of differences
 - Between epochs, receivers, transmitters
- No forming of linear combinations
 - Wide-lane, Ionosphere-free, ...
- Known influences are corrected beforehand
 - Relativistic effects, transmitter clock error, phase wind-up, ...
- Remaining influences are added as parameters
 - Ionospheric refraction, unknown ambiguities, antenna center variations, ...

Goal: kinematic orbits for gravity field determination

Observation equations

Code pseudo range

$$R_i^k - \delta R_i^k = \rho_i^k + c\delta t_i + I_i^k + ACV_i + ACV^k$$

known
influences

$$\Phi_i^k - \delta\Phi_i^k = \rho_i^k + c\delta t_i + \lambda n_i^k + I_i^k + ACV_i + ACV^k$$

Phase observation

Observation equations

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known
influences

geometrical
distance

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Phase observation

Observation equations

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known influences

geometrical distance

Receiver clock error

$$\Phi_i^k - \delta\Phi_i^k = \rho_i^k + c\delta t_i + \lambda n_i^k + I_i^k + ACV_i + ACV^k$$

Phase observation

Observation equations

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geometrical distance

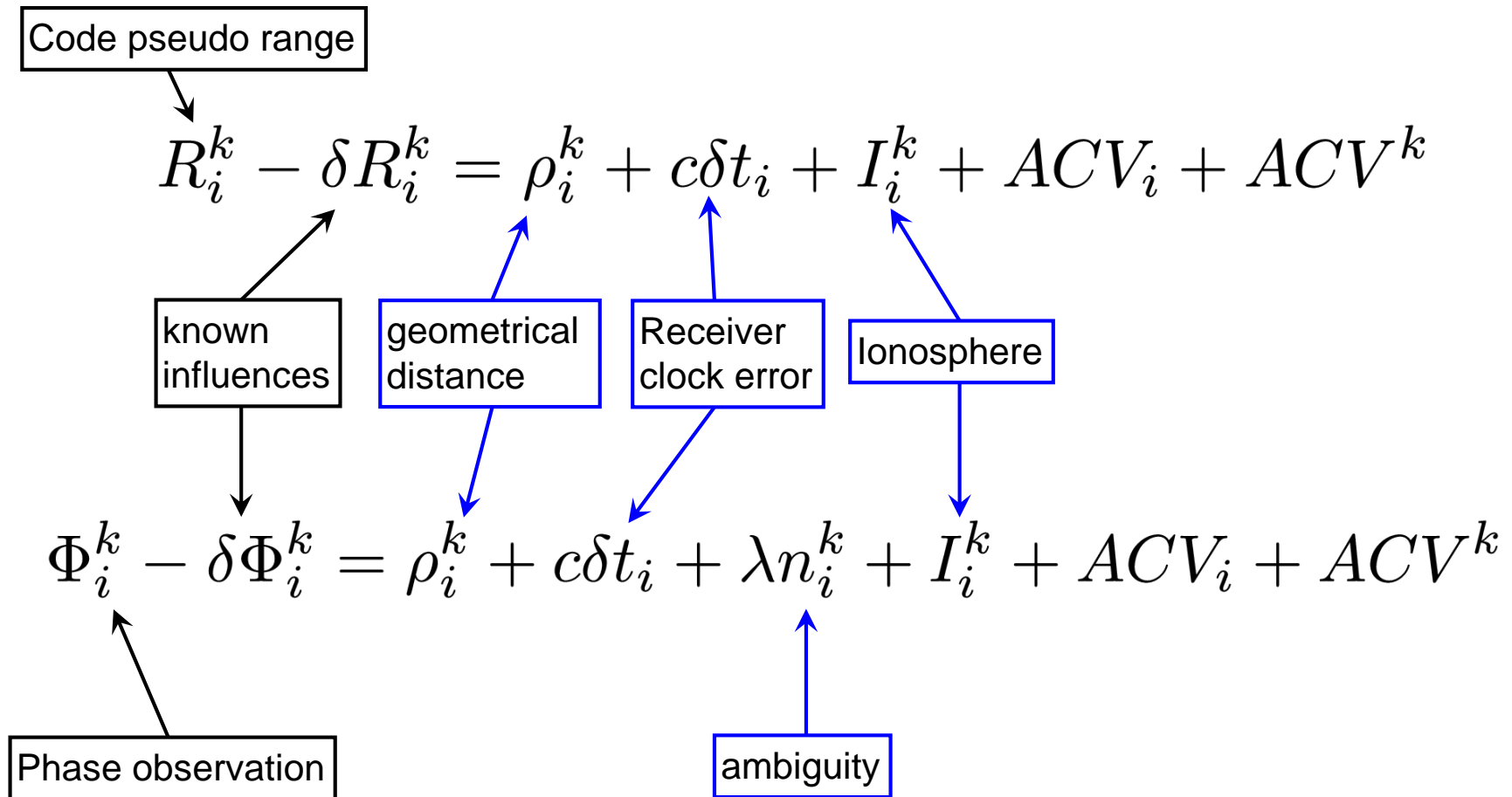
Receiver clock error

$$\Phi_i^k - \delta\Phi_i^k = \rho_i^k + c\delta t_i + \lambda n_i^k + I_i^k + ACV_i + ACV^k$$

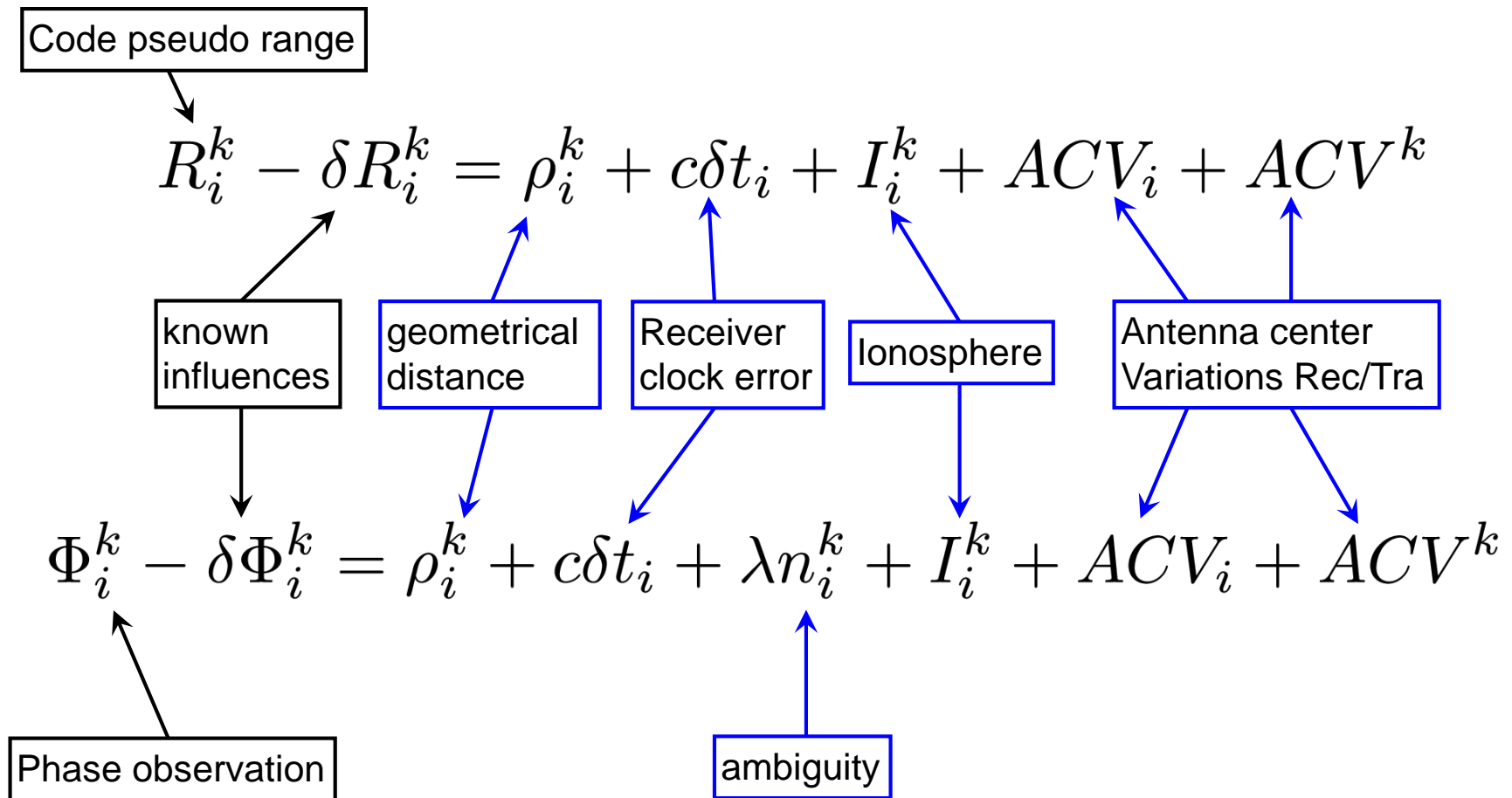
Phase observation

ambiguity

Observation equations



Observation equations



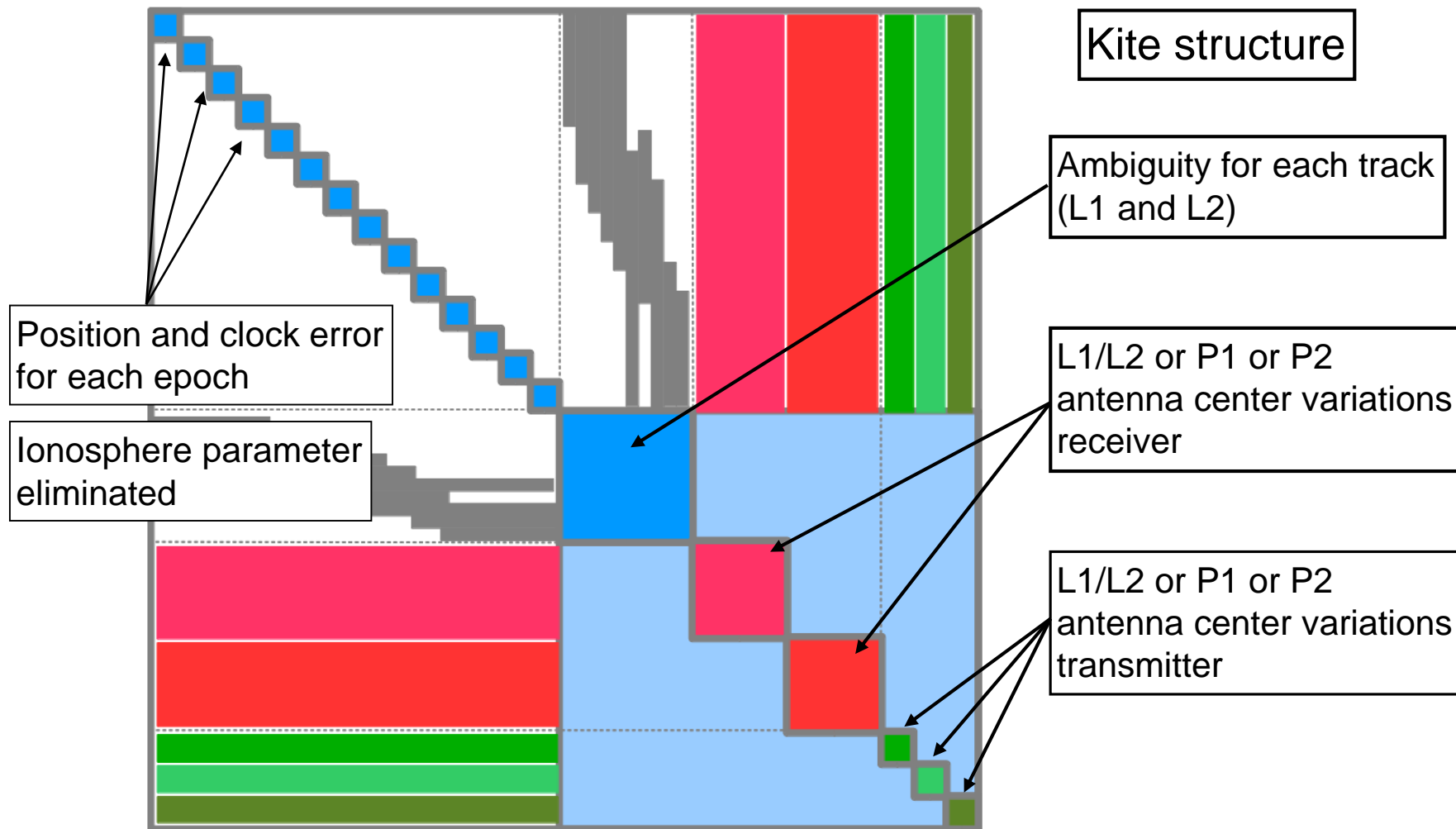
Building normal equation

- Epoch dependent parameters
 - Position (x,y,z), receiver clock error and ionosphere parameter for each satellite-receiver combination
 - Ionosphere parameter is eliminated
 - 4x4 block remains for each epoch (main diagonal)
- Epoch independent parameters
 - Ambiguities
 - for each continuous track of a satellite
 - for each carrier frequency
 - Antenna center variations
 - Receiver
 - Transmitter

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Normal equation



Antenna center variations

For code and phase observations

For receiver

- Spherical harmonics expansion

For transmitters

- Radial basis functions

Degree 0 and 1 omitted to avoid singularity
(constant and origin of antenna)

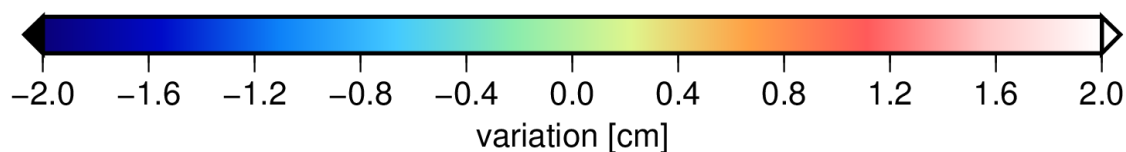
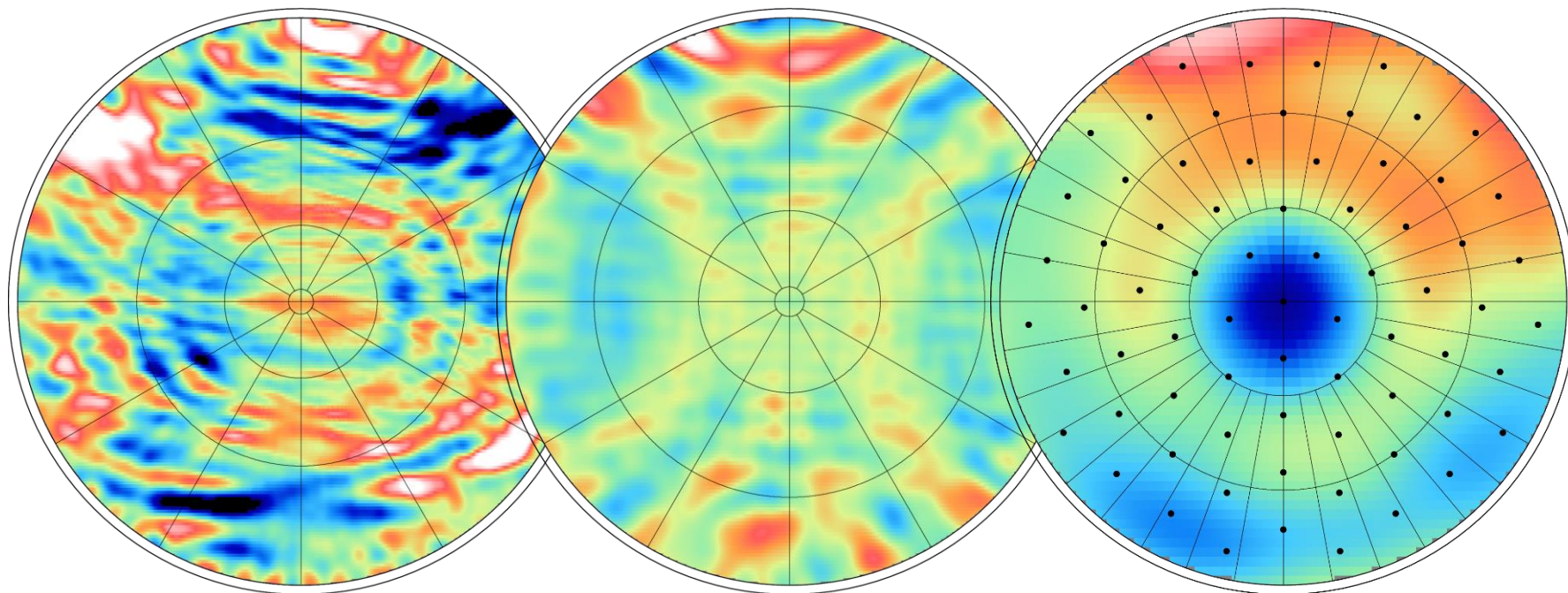
Both representations are azimuth-elevation dependent

Antenna center variations – Example L1/L2

• GOCE

• GRACE A

• SVN 41 (Block IIR-A)

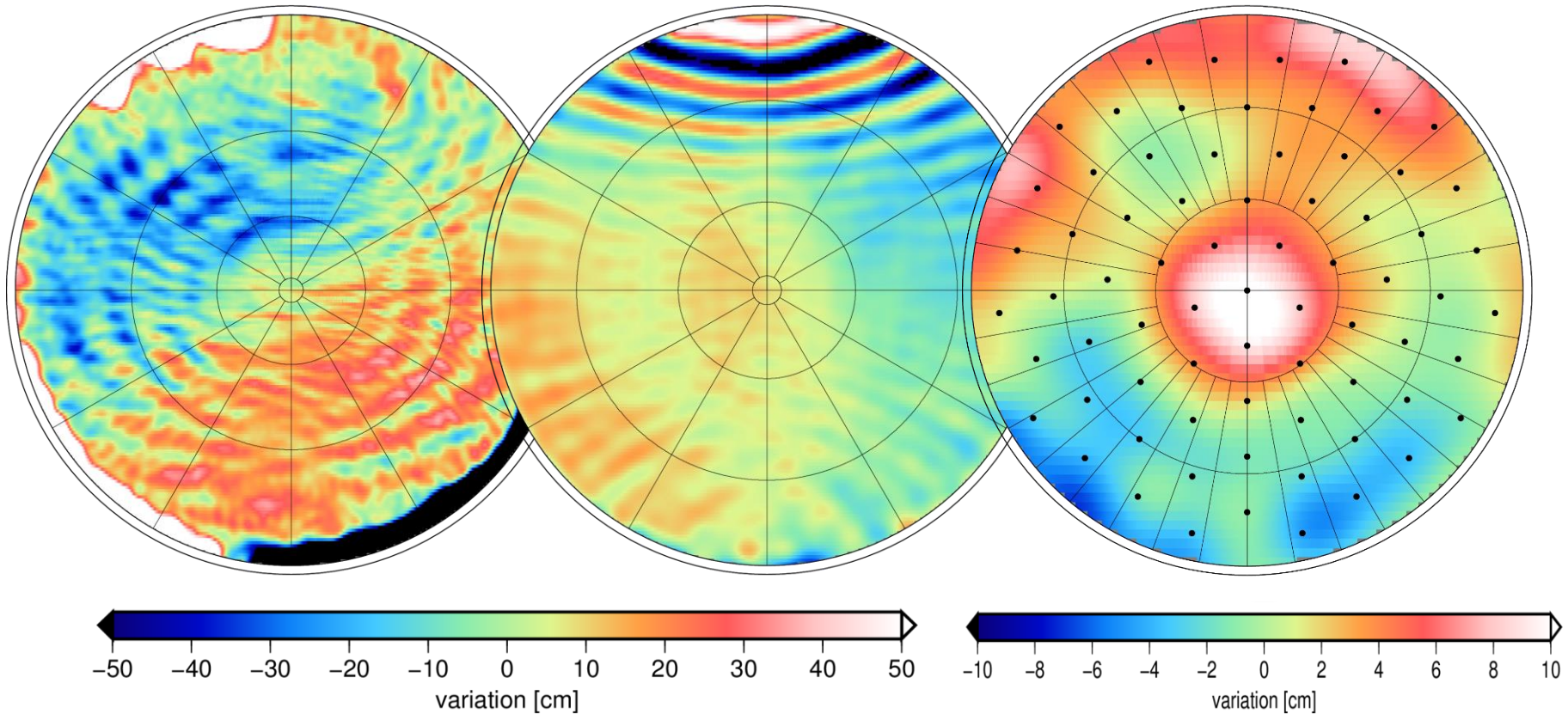


Antenna center variations – Example P1

• GOCE

• GRACE A

• SVN 41 (Block IIR-A)



Raw GNSS measurements

- Measurements as they are observed

Advantages

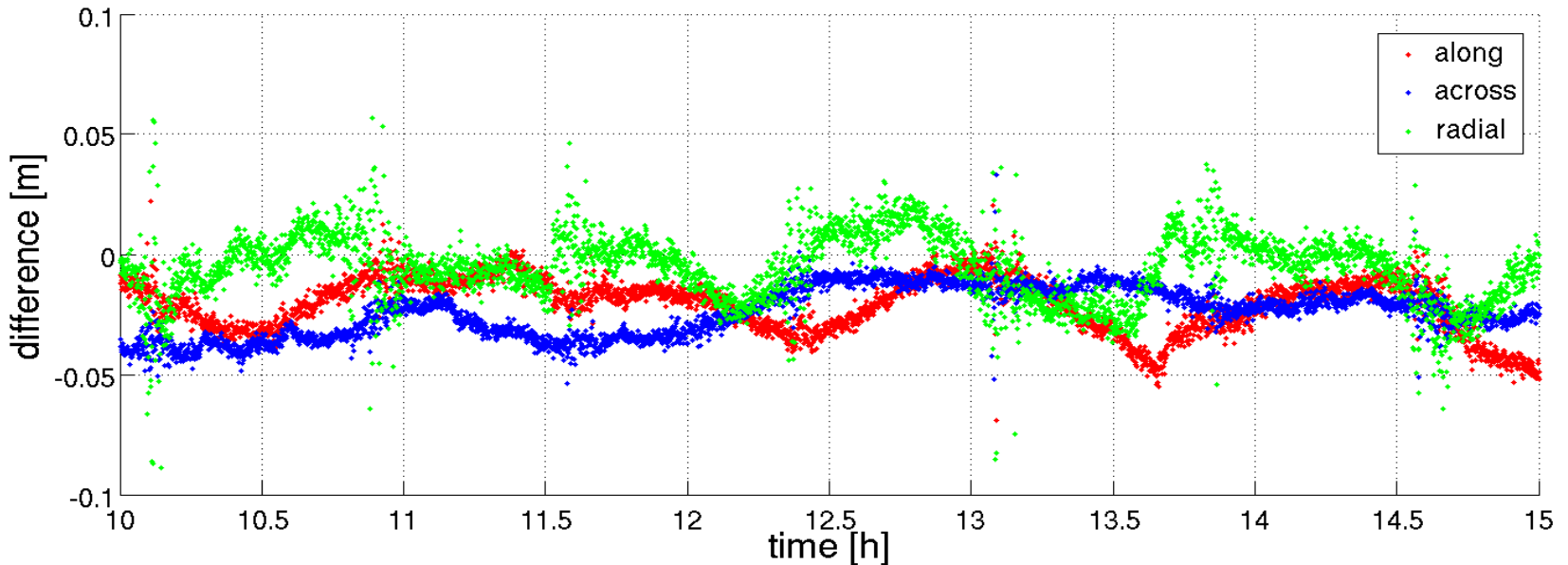
- Original noise level
- Parameters are not mixed up
 - Ambiguities on each frequency can be accessed directly
 - Gives possibility to fix them

Disadvantages

- Influences are not eliminated/reduced

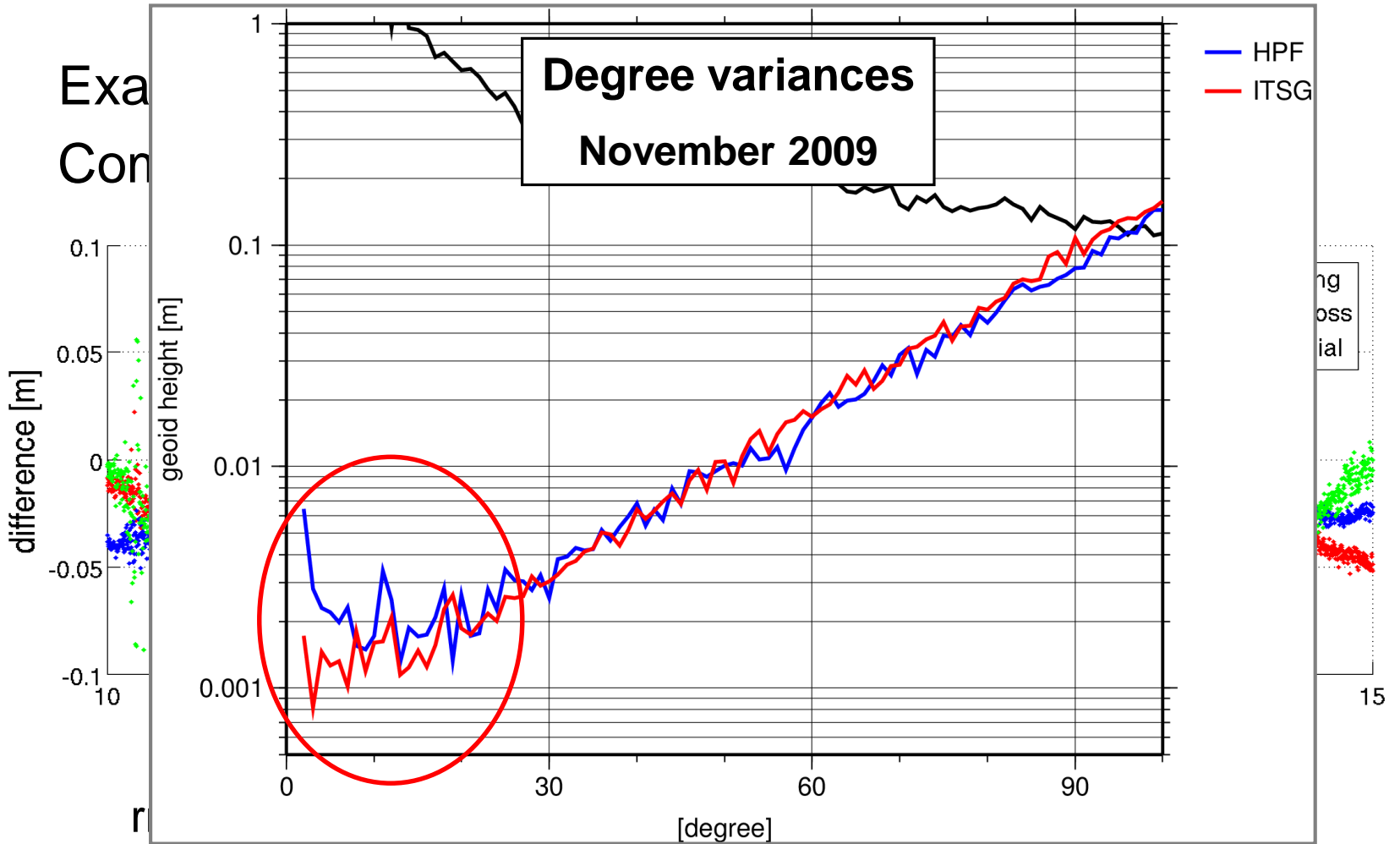
Results for GOCE

Example: GOCE 02.11.2009 10:00 – 15:00
Compared to official reduced-dynamic orbit



rms: along: 1.7 cm / across: 1.2 cm / radial: 2.3 cm

Results for GOCE



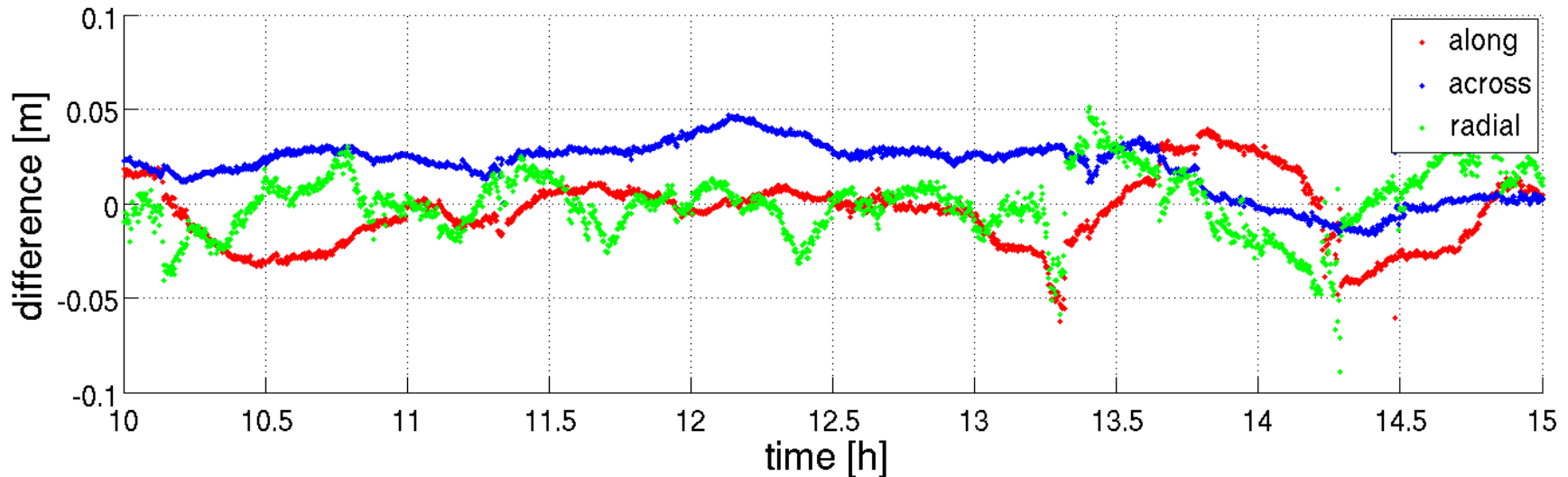
Results for GRACE

Example: GRACE A, 01.05.2008 10:00 – 15:00

Compared to official reduced-dynamic orbit

IGS values for transmitter antenna center variations used

⇒ nadir dependent



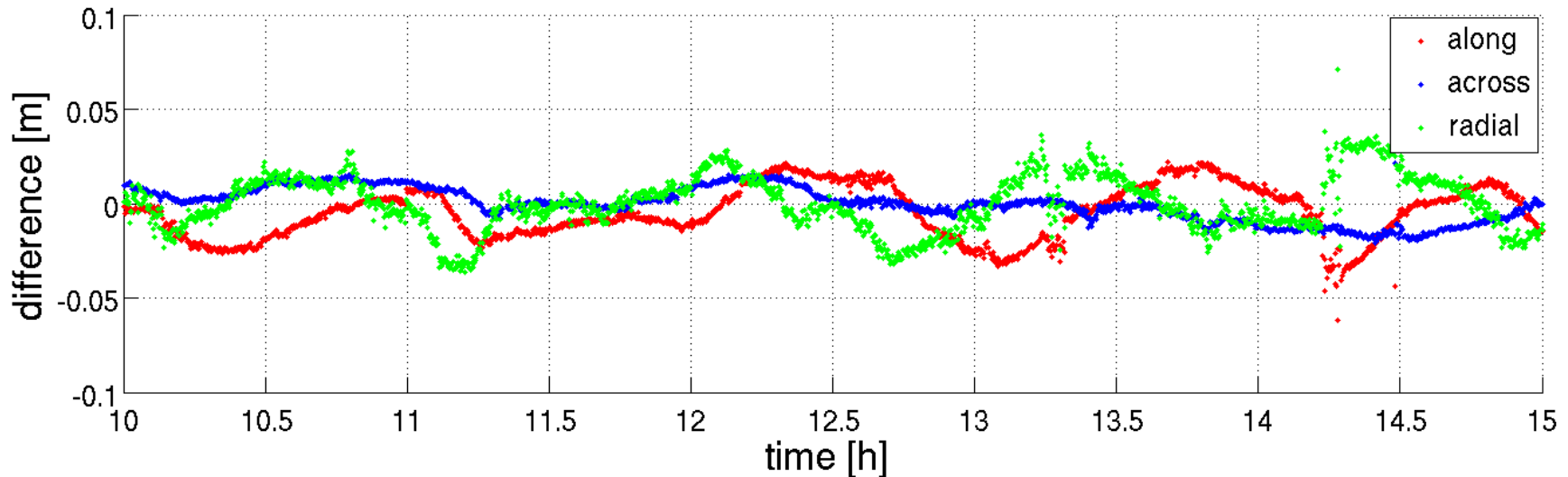
rms: along: 2.3 cm / across: 2.0 cm / radial: 2.7 cm

Results for GRACE

Example: GRACE A, 01.05.2008 10:00 – 15:00

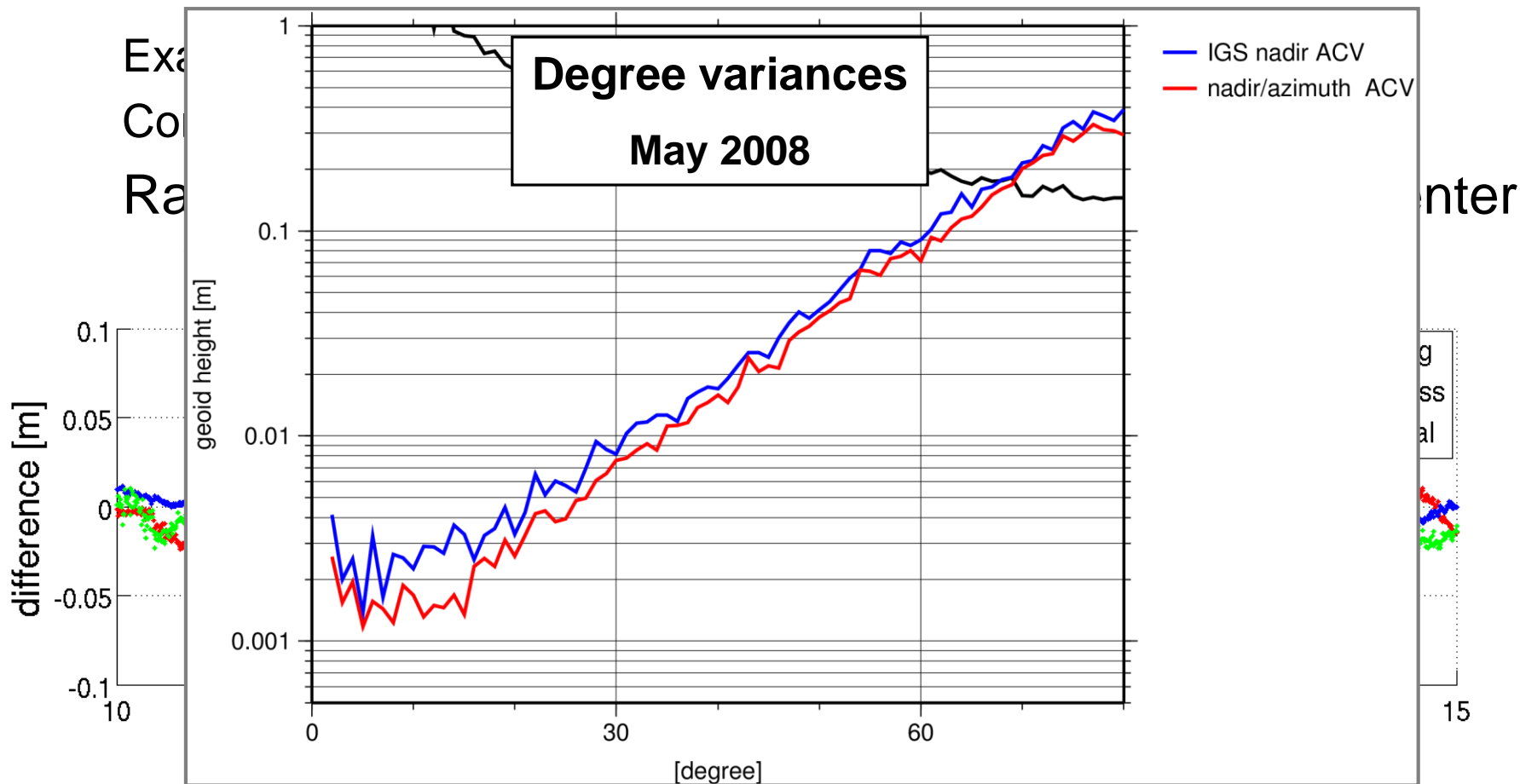
Compared to official reduced-dynamic orbit

Radial basis functions used for transmitter antenna center variations \Rightarrow azimuth-nadir dependent



rms: along: 1.8 cm / across: 1.5 cm / radial: 2.0 cm

Results for GRACE



Conclusions

- Method is straightforward
- Well suited for modernized GNSS environment with additional frequencies (L5,...)
- Ambiguities are directly accessible on each frequency \Rightarrow integer nature is preserved
- Antenna center variations can be estimated
 - For receivers and transmitters
 - For phase and code observations
- Current results are suitable for gravity field recovery

Conclusions

- Next steps:
 - Ambiguity fixing
 - Compute longer time series
 - CHAMP, GRACE, and GOCE
 - Prepare for SWARM
 - Validation in terms of position and gravity field results

- The orbits will be published: itsg.tugraz.at

Thank you for your attention!